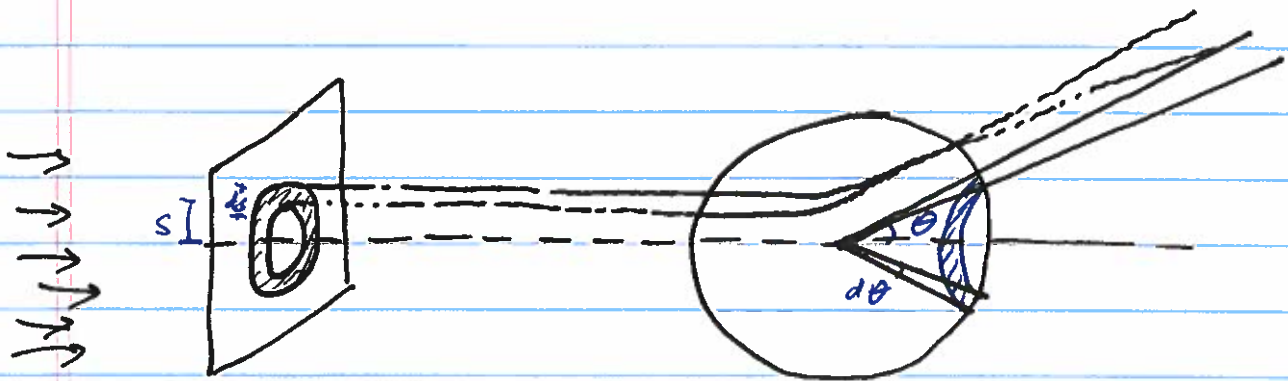


Review: Scattering, Rigid Body Motion, Fictitious Forces.

Scattering



Differential Scattering Cross section $\sigma(\Omega)$

$$\sigma(\Omega) d\Omega = \frac{\# \text{ particles scattered into solid angle } d\Omega / \text{time}}{\# \text{ particles in beam/unit area/unit time}}$$

$\sigma(\Omega)$ has units of area — heuristically measures effective size of scattering source.

Central force: $\sigma(\Omega)$ is symmetric about axis through scattering center: $d\Omega = 2\pi \sin\theta d\theta$

$$\sigma(\Omega) d\Omega = 2\pi \sin\theta \sigma(\theta) d\theta$$

$$\sigma(\theta) = \frac{s}{\sin\theta} \left| \frac{ds}{d\theta} \right|$$

Impact parameter
 \uparrow $s =$ function of beam energy E , and θ .

(Really, $\theta =$ function of E, s)

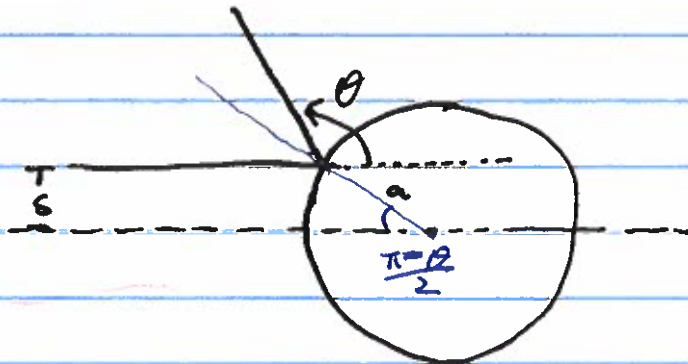
Example: Rutherford Cross Section

Analysis of trajectories in potential $V = \frac{ZZ'e^2}{r}$

$$\text{gives } S = \frac{ZZ'e^2}{2E} \cot\left(\frac{\theta}{2}\right)$$

$$\sigma(\theta) = \frac{1}{4} \left(\frac{ZZ'e^2}{2E} \right)^2 \csc^4\left(\frac{\theta}{2}\right)$$

Example: Hard sphere scattering, sphere of radius a



$$S = a \sin\left(\frac{\pi - \theta}{2}\right) = a \cos \frac{\theta}{2}$$

$$\sigma(\theta) = \frac{S}{\sin \theta} \left| \frac{dS}{d\theta} \right| = \frac{a \cos \theta/2}{\sin \theta} \cdot \frac{a}{2} \sin \frac{\theta}{2}$$

$$= \frac{a^2}{4}$$

$$\sigma = \int_0^\pi \sigma(\theta) \cdot 2\pi \sin \theta \, d\theta = -2\pi \frac{a^2}{4} \cos \theta \Big|_0^\pi$$

$$= \pi a^2$$

= cross-sectional area of sphere.

Orthogonal Transformations

If the components of a vector \vec{v} transform as
 $\vec{v}' = \underline{A} \vec{v}$ for some 3×3 matrix \underline{A} ,
then $|\vec{v}'| = |\vec{v}|$ for all \vec{v} if

$$(\underline{A}\vec{v})^T (\underline{A}\vec{v}) = \vec{v}^T \underline{A}^T \underline{A} \vec{v} = \vec{v}^T \vec{v}$$

$$\rightarrow \boxed{\underline{A}^T \underline{A} = \underline{1}}, \text{ or } \underline{A}^T = \underline{A}^{-1}$$

$\det \underline{A} = +1 \rightarrow$ Proper orthogonal transformations

$\det \underline{A} = -1 \rightarrow$ Improper orthogonal transformations —
not smoothly connected to $\underline{1}$.

Eigenvalues of proper orthogonal \underline{A} are $1, e^{i\Phi}, e^{-i\Phi}$

By a similarity transformation, \underline{A} can be put in the form,

$$\underline{S}^{-1} \underline{A} \underline{S} = \begin{pmatrix} \cos \Phi & \sin \Phi & 0 \\ -\sin \Phi & \cos \Phi & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

This is a rotation about the z -axis.

Euler's Theorem: The general displacement of a rigid body
with one point fixed is a rotation about some axis.

Euler Angles - Describe orientation by 3 successive rotations from space axes to body axes

- 1) Rotate about z-axis by ϕ
- 2) Rotate about new x-axis by θ
- 3) Rotate about new z-axis by ψ

Multiplying the three rotation matrices gives:

$$\underline{A} = \begin{pmatrix} C_{\psi} C_{\phi} - C_{\theta} S_{\phi} S_{\psi} & C_{\psi} S_{\phi} + C_{\theta} C_{\phi} S_{\psi} & S_{\psi} S_{\theta} \\ -S_{\psi} C_{\phi} - C_{\theta} S_{\phi} C_{\psi} & -S_{\psi} S_{\phi} + C_{\theta} C_{\phi} C_{\psi} & C_{\psi} S_{\theta} \\ S_{\theta} S_{\phi} & -S_{\theta} C_{\phi} & C_{\theta} \end{pmatrix}$$

where $C_{\psi} = \cos \psi$, $S_{\psi} = \sin \psi$, etc.

$\underline{\vec{x}}' = \underline{A} \underline{\vec{x}}$, where $\underline{\vec{x}}'$ is expressed in the body frame, $\underline{\vec{x}}$ is expressed in the space frame.

To convert from body frame to space frame, transpose \underline{A} :

$$\underline{\vec{x}} = \underline{A}^{-1} \underline{\vec{x}}' = \underline{A}^T \underline{\vec{x}}' \quad \text{because } \underline{A} \text{ is orthogonal.}$$

Rate of change of a vector

$\left(\frac{d\vec{a}}{dt}\right)_{\text{space}}$ — rate of change of \vec{a} relative to space frame (inertial)

$\left(\frac{d\vec{a}}{dt}\right)_{\text{body}}$ — rate of change of \vec{a} relative to body frame, fixed relative to body (non-inertial)

$$\boxed{\left(\frac{d\vec{a}}{dt}\right)_{\text{space}} = \left(\frac{d\vec{a}}{dt}\right)_{\text{body}} + \vec{\omega} \times \vec{a}}$$

instantaneous angular velocity of body relative to space.

In the body frame, in terms of the Euler angles,

$$\vec{\omega} = \begin{pmatrix} \dot{\phi} \sin\theta \sin\psi + \dot{\psi} \cos\psi \\ \dot{\phi} \sin\theta \cos\psi - \dot{\psi} \sin\psi \\ \dot{\phi} \cos\theta + \dot{\psi} \end{pmatrix}$$

In the space frame,

$$\vec{\omega} = \begin{pmatrix} \dot{\psi} \cos\psi + \dot{\phi} \sin\theta \sin\phi \\ \dot{\psi} \sin\psi - \dot{\phi} \sin\theta \cos\phi \\ \dot{\psi} \cos\theta + \dot{\phi} \end{pmatrix}$$

Coriolis and Centrifugal Effects (Fictitious forces)

Consider a uniformly rotating frame w/ angular velocity $\vec{\omega}$ relative to the inertial space frame.

$$\vec{v}_s = \left(\frac{d\vec{r}}{dt} \right)_s = \left(\frac{d\vec{r}}{dt} \right)_r + \vec{\omega} \times \vec{r} = \vec{v}_r + \vec{\omega} \times \vec{r}$$

$$\begin{aligned} \vec{a}_s &= \left(\frac{d^2\vec{r}}{dt^2} \right)_s = \frac{d}{dt} (\vec{v}_r + \vec{\omega} \times \vec{r})_r + \vec{\omega} \times (\vec{v}_r + \vec{\omega} \times \vec{r}) \\ &= \left(\frac{d\vec{v}_r}{dt} \right)_r + 2(\vec{\omega} \times \vec{v}_r) + \vec{\omega} \times (\vec{\omega} \times \vec{r}) \end{aligned}$$

$$\equiv \vec{a}_r + 2(\vec{\omega} \times \vec{v}_r) + \vec{\omega} \times (\vec{\omega} \times \vec{r})$$

$$\vec{p} = m\vec{a}_s \Rightarrow m\vec{a}_r = \vec{F} - \underbrace{2m(\vec{\omega} \times \vec{v}_r)}_{\text{Coriolis effect}} - \underbrace{m\vec{\omega} \times (\vec{\omega} \times \vec{r})}_{\text{Centrifugal effect}}$$

We may use Newton's 2nd Law in the rotating (non-inertial) frame if we add fictitious forces proportional to m to the external force \vec{F} .

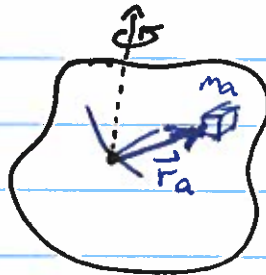
Inertia Tensor

$$\vec{L} = \underline{I} \vec{\omega}, \text{ or } L_i = I_{ij} \omega_j$$

angular momentum Inertia tensor angular velocity

$$I_{ij} = \sum_a m_a (r_a^2 \delta_{ij} - r_{ai} r_{aj})$$

Inertia tensor for motion w/ one point (the origin) fixed.



Continuous medium:

$$I_{ij} = \int dV \rho(\vec{r}) (r^2 \delta_{ij} - x_i x_j)$$

Under an orthogonal transformation:

$$\underline{I}' = \underline{A} \underline{I} \underline{A}^T, \text{ or } I'_{ij} = a_{ik} a_{jl} I_{kl}$$

- transformation law for rank-2 tensor under rotations.

Kinetic Energy: $T = \frac{1}{2} \vec{\omega} \cdot \underline{I} \vec{\omega}$

Define $\vec{w} = \omega \hat{n}$.

$$T = \frac{1}{2} (\hat{n} \cdot \underline{I} \hat{n}) \omega^2 \equiv \frac{1}{2} I \omega^2$$

$I = \hat{n} \cdot \underline{I} \hat{n}$ = moment of inertia about axis of rotation.

Principal Axes and Principal Moments of Inertia

\underline{I} can be diagonalized by a similarity transformation by an orthogonal matrix A :

$$\underline{I}_D = \begin{pmatrix} I_1 & 0 & 0 \\ 0 & I_2 & 0 \\ 0 & 0 & I_3 \end{pmatrix} = \underline{A} \underline{I} \underline{A}^T \text{ for some orthogonal } \underline{A}.$$

I_1, I_2, I_3 — principal moments of inertia
= eigenvalues of \underline{I} .

Principal axes = eigenvectors of \underline{I} .

In principal axis frame, $\vec{L} = \underline{I} \vec{\omega}$ becomes

$$L_1 = I_1 \omega_1, \quad L_2 = I_2 \omega_2, \quad L_3 = I_3 \omega_3$$

Euler's Equations of Motion:

In an inertial frame w/ origin at the fixed pt. of the rigid body,

$$\left(\frac{d\vec{L}}{dt}\right)_S = \vec{N} \quad \text{torque}$$

In ^{terms of} the body frame, $\left(\frac{d\vec{L}}{dt}\right)_S = \left(\frac{d\vec{L}}{dt}\right)_B + \vec{\omega} \times \vec{L}$

Choose the principal axis frame in the body, so that $L_i = I_i \omega_i$, etc.

$$\Rightarrow \left. \begin{aligned} I_1 \dot{\omega}_1 - \omega_2 \omega_3 (I_2 - I_3) &= N_1 \\ I_2 \dot{\omega}_2 - \omega_3 \omega_1 (I_3 - I_1) &= N_2 \\ I_3 \dot{\omega}_3 - \omega_1 \omega_2 (I_1 - I_2) &= N_3 \end{aligned} \right\} \text{Euler's equations of motion.}$$

Torque-free motion: $N_1 = N_2 = N_3 = 0$.

Symmetric body: suppose $I_1 = I_2$. Then $\dot{\omega}_3 = 0$.

- Motion generally precesses about ω_3 axis in body frame.

with torque: Motion is more complicated.

Symmetric top w/ $I_1 = I_2$, torque due to gravity

- precession + nutation (bobbing) of figure axis.